

Powerline Communication for Spacecraft and Instruments

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> PLC technology presents an opportunity to decrease mass and complexity of spacecraft and GSE harnessing

Project Objective: Enable simpler and more capable spacecraft through elimination of data harnessing with powerline communication (PLC)

- Survey, evaluate, and document power bus noise and the frequency spectrum of PLC technologies
- Demonstrate PLC functionality through a spacecraft to instrument like



- interface
- Characterize the robustness of the PLC interface through EMI/EMC and environmental testing



PLC Prototype/Daughter Card PWB



PLC Prototype System Block Diagram

Benefits to NASA and JPL:

Reduction in spacecraft harnessing mass and

PLC Prototype Thermal Ambient Test Setup

FY18/19 Results:

Primary objectives of FY19 were the design, fabrication, and test of a custom JPL PLC Prototype module using a radiation tolerant commercial off the shelf (COTS) PLC modem. Basing the design around the MAX79356 PLC modem, a set of PLC Prototype boards was created and both functional and environmental testing carried out. Full-duplex (bidirectional) communication across a standard 28V spacecraft bus was achieved with a maximum data rate of 180kbps. No loss of data rate was observed during thermal ambient testing between -35C and 55C, closely matching typical allowable flight temperatures (AFT) for JPL missions.

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- telemetry/command interface complexity
- Creates simpler instrument/spacecraft interfaces, more straightforward instrument packaging, and the capability for more deployable instruments
- With PLC the NuSTAR mission conductor count between spacecraft and instrument could have been reduced by 32.6%, resulting in a mass reduction of 1.36kg
- This reduction in mass provides the capability to add additional instruments or fuel to the spacecraft, enabling greater science and mission possibilities.



CE01/03 28V High Side Power Line

RE02, Vert. Pol. 14kHz – 30MHz







