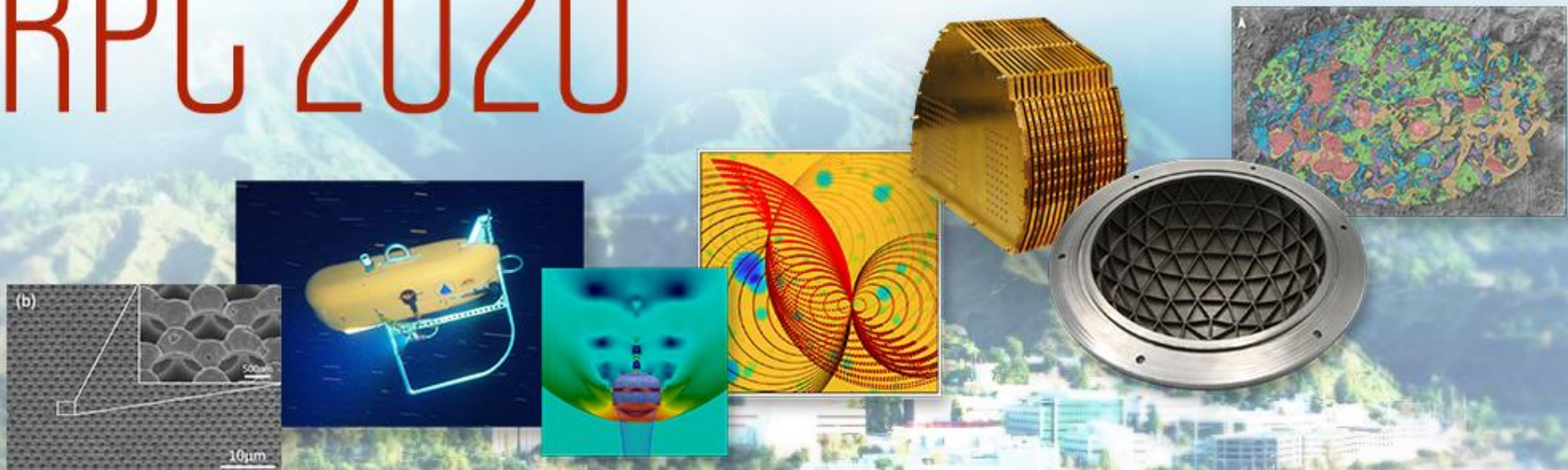


# RPC 2020



## Virtual Research Presentation Conference

Light-Driven Electrochemical Production of Oxygen and Fuel from CO<sub>2</sub> and Sunlight for Mars ISRU

**Principal Investigator: John-Paul Jones (346)**

**Co-Is: Keith Billings (346), Jillian Clinton (Caltech), Harry Gray (Caltech), Emmanuelle Despagnet-Ayoub (346)**

**Program: Strategic Initiative**

Assigned Presentation # RPC-151

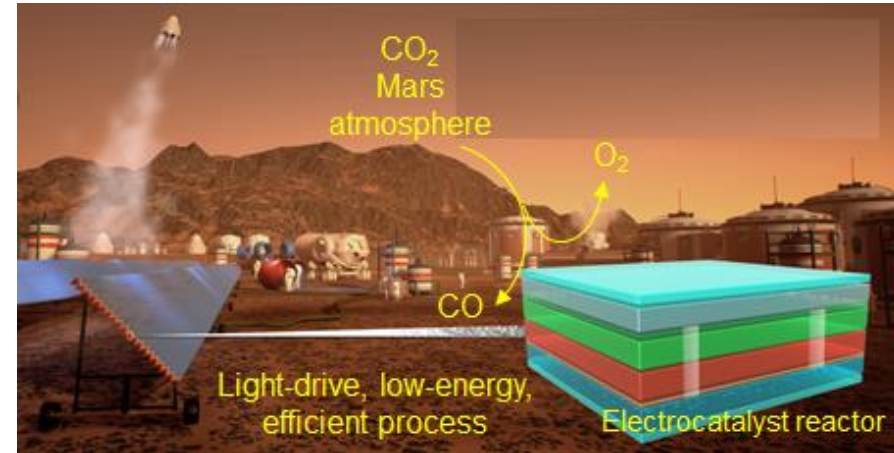


**Jet Propulsion Laboratory**  
California Institute of Technology



# Oxygen generation on Mars is a critical technology

- Future human Mars missions
  - Preliminary estimates of 30,000 kg of O<sub>2</sub> for Martian ascent vehicle<sup>1</sup>
- Human presence
  - Each human exhales ~1 kg CO<sub>2</sub>/day<sup>2</sup>
- Electrochemical reduction reactions:
  - Cathode:  $\text{CO}_2 + 2\text{H}^+ + 2\text{e}^- \rightarrow \text{CO} + \text{H}_2\text{O}$
  - Anode:  $\text{H}_2\text{O} \rightarrow 2\text{H}^+ + 2\text{e}^- + \text{O}_2$



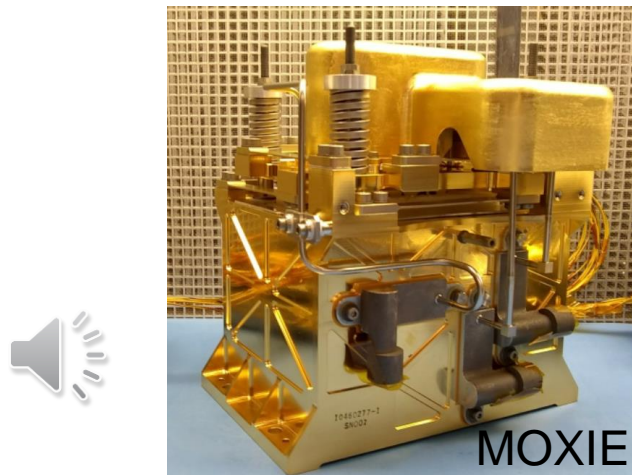
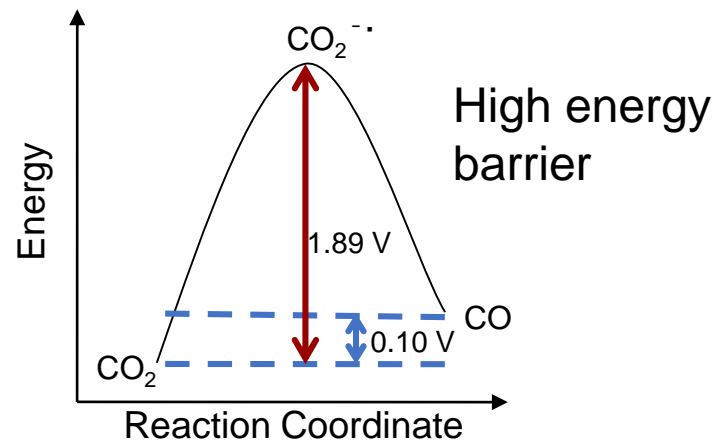
**The goal of this project is to develop a low temperature O<sub>2</sub> generation process powered by sunlight**

1. NASA's Human Exploration Design Reference Architecture 5.0 Addendum
2. D. E. Sherif and J. C. Knox, "International Space Station Carbon Dioxide Removal Assembly (ISS CDRA) Concepts and Advancements" *SAE Tech. Pap.* 2005-01-2892 (2005).

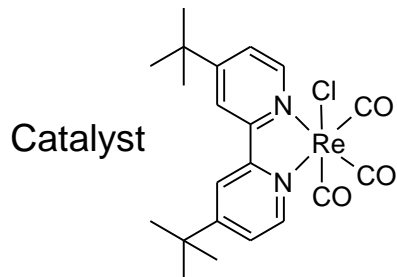


## Problem Description

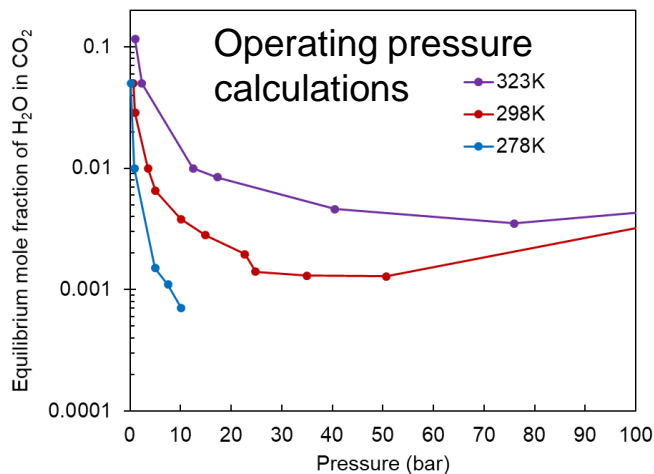
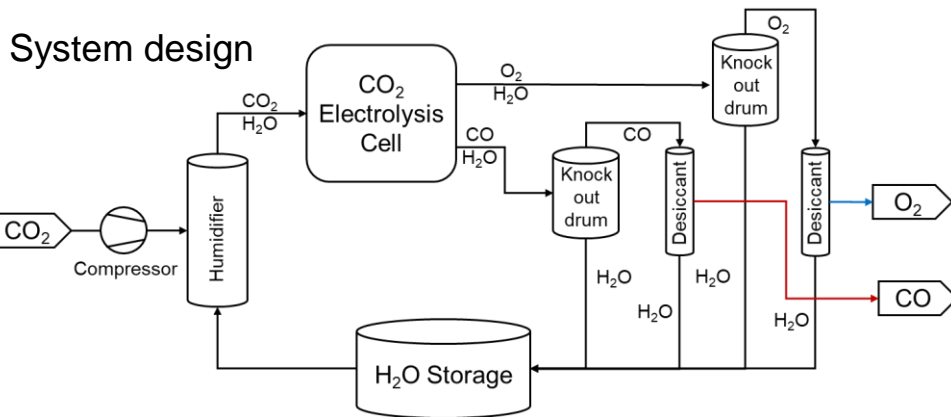
- CO<sub>2</sub> reduction is very challenging due to high energy barrier
- SOA: Solid Oxide Electrolysis Cell (SOEC) at 800 °C
- Oxide membrane enables conversion of CO<sub>2</sub> to CO and O<sub>2</sub> without protons
  - Cathode:  $2 \text{CO}_2 + 4 \text{e}^- \rightarrow 2 \text{CO} + 2 \text{O}^{2-}$
  - Anode:  $2 \text{O}^{2-} \rightarrow \text{O}_2 + 4 \text{e}^-$
- High temperature allows endothermic reaction
  - Some energy for conversion is provided by heat, rather than electrochemically
- Future Mars sample return missions and long term exploration will require access to O<sub>2</sub> for propellant and life support systems
- The Martian atmosphere provides an almost limitless supply of CO<sub>2</sub>, which can be electrochemically converted into O<sub>2</sub>



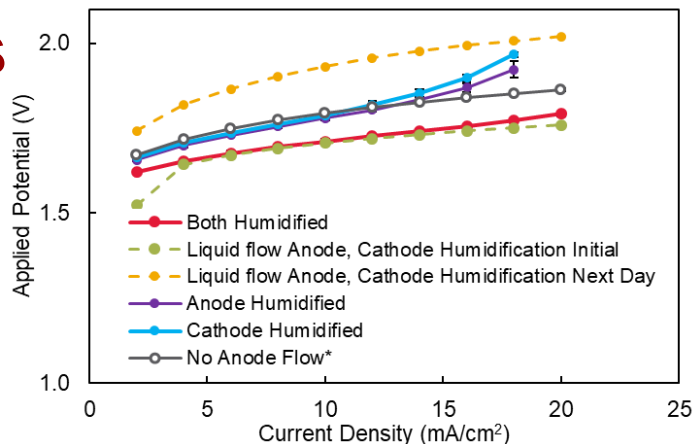
# Methodology



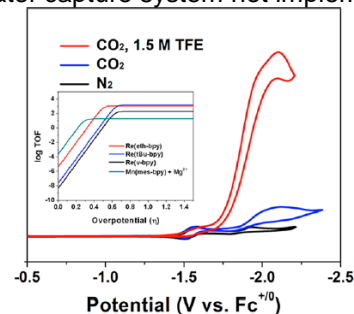
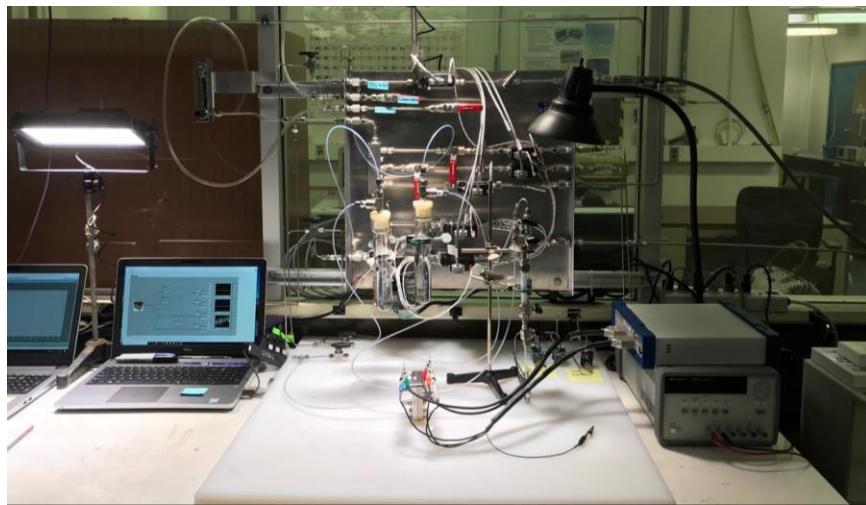
1. Develop efficient catalyst for CO<sub>2</sub> reduction
2. Integrate catalyst into flow electrolysis cell
3. Design an O<sub>2</sub> production system
4. Build a breadboard
5. Test electrochemical performance of system
6. Measure water loss, flow rate, pressure, temperature, system stability and other performance parameters for full system comparison to MOXIE



## Results



- Molecular CO<sub>2</sub> catalyst developed
- Catalyst integrated into flow cell
- End-to-end electrochemical CO<sub>2</sub> reduction system designed
- Breadboard system built
- Electrochemical performance evaluated
- Many system-level performance metrics measured
  - Flow rate effects measured
  - Humidification effects measured
  - Long term catalyst stability partially measured
  - Temperature and pressure not investigated
  - Water capture system not implemented



# Publications and References

## References:

1. NASA's Human Exploration Design Reference Architecture 5.0 Addendum
2. D. E. Sherif and J. C. Knox, "International Space Station Carbon Dioxide Removal Assembly (ISS CDRA) Concepts and Advancements" *SAE Tech. Pap.* 2005-01-2892 (2005).

## Publications:

1. Almagul Zhanaidarova, Curtis E. Moore, Milan Gembicky and Clifford P. Kubiak, *Chemical Communications*, 2018, **54**, 4116-4119.
2. Almagul Zhanaidarova, Andrew L. Ostericher, Christopher J. Miller, Simon C. Jones and Clifford P. Kubiak, *Organometallics*, 2019, **38**, 1204-1207.