

# RPC 2020



## Virtual Research Presentation Conference

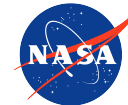
### Microarcsecond Astrometry Telescope Instrument on Small Satellite

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**Program: Topical R&TD**

Assigned Presentation #RPC-102



**Jet Propulsion Laboratory**  
California Institute of Technology



## Introduction

### Abstract

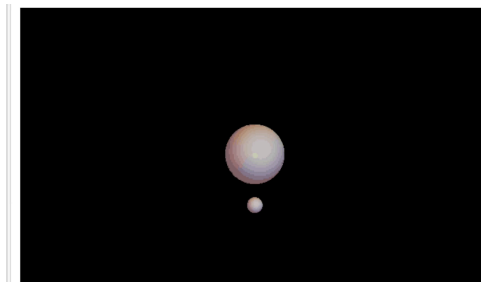
"MASS" (Micro-arcsecond Astrometry Small Satellite) is an astrophysics mission designed to produce the first survey of the habitable zone (HZ) planets in the nearby stellar neighborhood. MASS is based on differential astrometry (angle measurements) of a target star relative to a background of reference stars.

While both radial-velocity and transit techniques have been successful at finding planets more generally, they have been ineffective at reaching Earth-mass planets in the HZ of nearby Sun-like stars due to the inherent constraints of these methods. Our astrometric approach overcomes these limitations.

This mission idea is based on the key focal plane calibration technique previously demonstrated in our laboratory using 80x80 pixel camera. The new MASS mission concept requires a large size CMOS detector (~150Mpx) calibration to achieve the micro-arcsecond level accuracy.



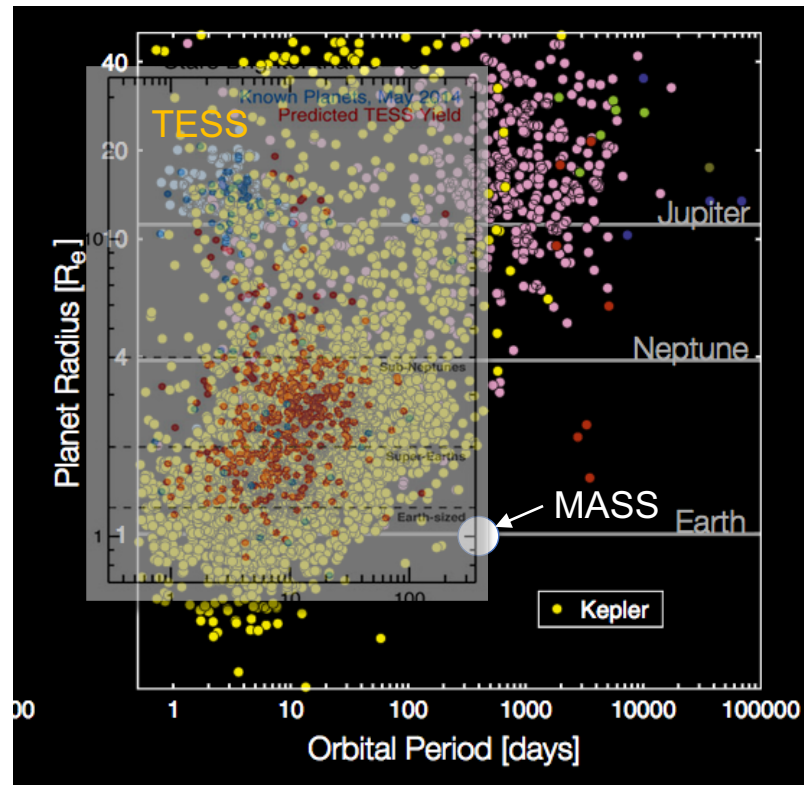
- **Transit method** can NOT detect non-transit planet
- **RV method** will have " $M \cdot \sin(i)$ " ambiguity
- **Astrometry method** (wobble) will detect both non-transit/transit, and enhance planet mass estimation





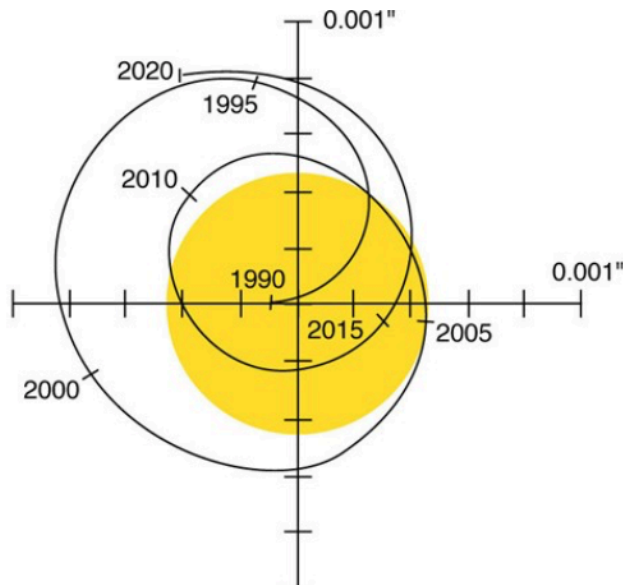
## Introduction

- KEPLER and TESS
  - Will not search an Earth like planet around G stars in 1 year orbit
  - Not astrometry mission (transit methods)
- MASS will be the First Astrometry mission, designed to search for Earth like planets around nearby 20 FGK stars
  - 1 AU, 1-3 Earth Masses
  - Thousands of observations / three year
  - Achieve 2 orders of magnitude improvement in astrometric precision over ESA's Gaia mission.

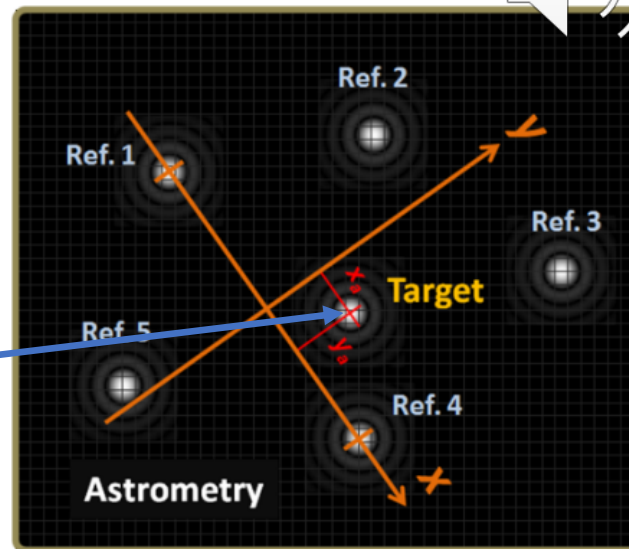




## Differential Astrometry



Solar wobble example

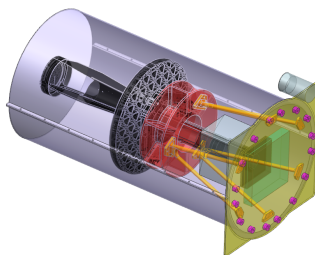


- MASS is a targeted-astrometry mission. It will measure the position of a target star relative to  $\sim 100$  reference stars in the same field. This is “relative” astrometry, as compared to *Gaia*’s “absolute” astrometry.
- MASS will not measure the *absolute position* of the target star at the  $\mu\text{as}$ -level but rather measure the *motion* of the target star at the  $\mu\text{as}$ -level.

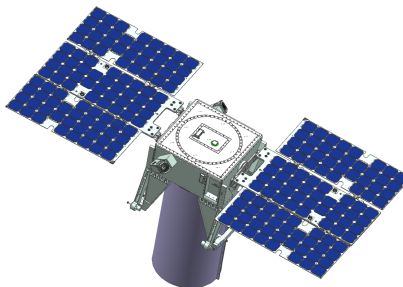


## MASS Overview

- Goal is  $\sim 4\mu\text{s}$  (1hr) astrometric precision
- Search for Earth mass planets  
Around  $\sim 20$  nearest FGK stars  
  
 $\sim$ five of 1M and  $\sim$ twelve of 2M  
Planets in the mid Habitable Zone
- Low cost mission taking advantages of low cost commercial spacecraft, SiC telescope



- A commercial, 35 cm aperture, SiC corrected RC telescope
- $6\mu\text{s}$  accuracy over 30 min observation
- Detector calibration pre-launch

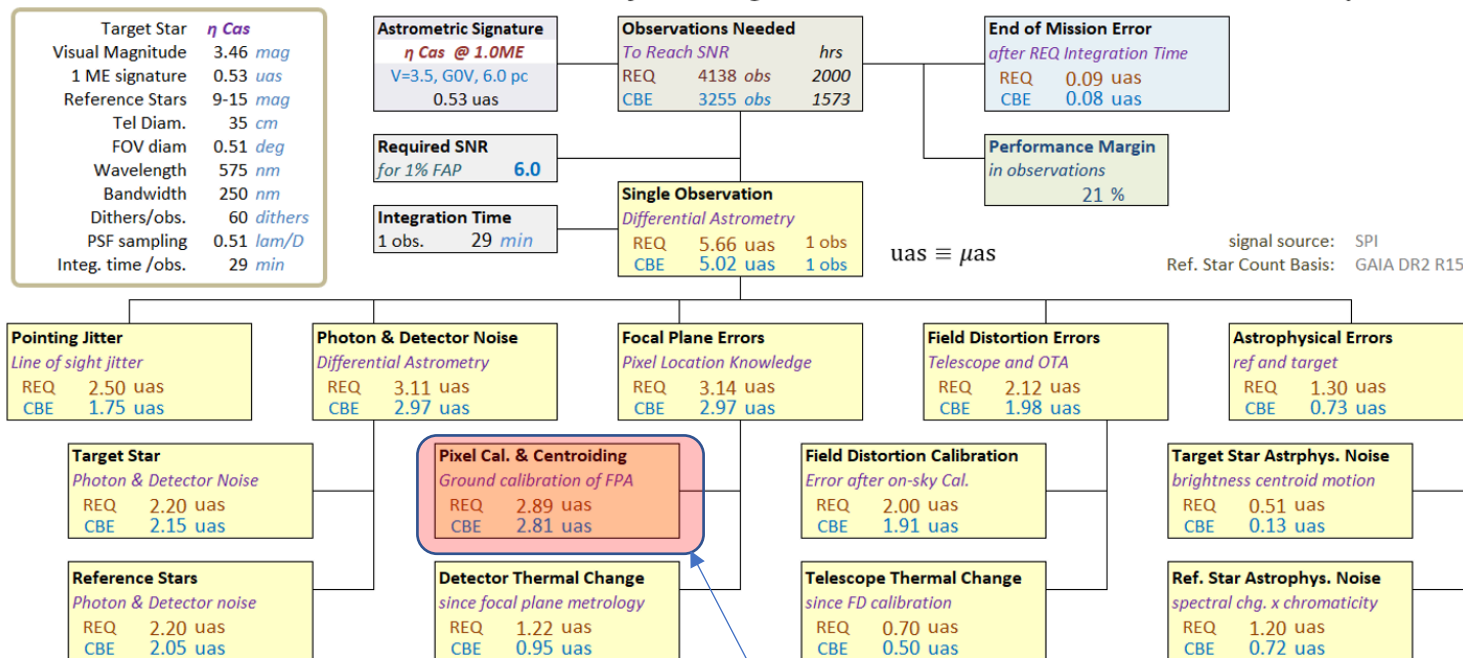


- 2x 200W solar panels
- 110 kg total mass
- ESPA-Grande compatible
- Propulsion for station-keeping

HIP	Name	Depth, ME	V mag	Spect. Type	Dist., pc	signature $\mu\text{s}$	Ref Stars	hours to SNR=6	cumul. hours
71683	$\alpha$ Cen A	1	0.0	G2V	1.3	2.42	1228	59	59
71681	$\alpha$ Cen B	1	1.4	K1V	1.3	1.71	1228	121	180
2021	$\beta$ Hyi	1	2.8	G2IV	7.5	0.55	105	957	1136
3821	$\eta$ Cas	1	3.5	G0V	6.0	0.53	488	1530	2667
77952	$\beta$ TrA	1	2.8	F1V	12.3	0.44	999	1511	4178
99240	$\delta$ Pav	2	3.6	G8IV	6.1	0.99	119	444	4622
22449	$\pi$ 3 Ori	2	3.2	F6V	8.0	0.98	139	543	5164
27072	$\gamma$ Lep	2	3.6	F6V	9.0	0.84	127	602	5766
746	$\beta$ Cas	2	2.3	F2III	16.7	0.87	372	703	6469
96100	$\sigma$ Dra	2	4.7	K0V	5.8	0.79	133	1236	7705
14632	$\iota$ Per	2	4.1	G0V	10.5	0.69	231	1377	9081
12777	$\theta$ Per	2	4.1	F8V	11.2	0.67	328	1591	10673
19849	40 Eri	2	4.4	K1V	5.0	0.89	77	1652	12325
105858	$\gamma$ Pav	2	4.2	F9V	9.2	0.72	94	1701	14026
8102	$\tau$ Ceti	2	3.5	G8V	3.6	1.31	28	1715	15741
108870	e Ind	2	4.7	K5V	3.6	0.96	65	1950	17691
1599	$\zeta$ Tuc	3	4.2	G0V	8.6	1.10	68	1238	18929
78072	$\gamma$ Ser	3	3.9	F6V	11.1	1.07	62	1340	20269
57757	$\beta$ Vir	3	3.6	F9V	10.9	1.14	41	1453	21722
64924	61 Vir	3	4.7	G7V	8.5	0.97	121	1661	23383
15510	e Eri	3	4.3	G6V	6.1	1.28	51	1961	25344
64394	$\beta$ Com	3	4.2	G0V	9.2	1.06	31	3419	28763



## MASS Error budget and key calibration requirement



- Pixel irregularities of the detector have to be calibrated to one part in 10,000.
- Laser interferometer fringes are used to calibrate pixel positions.



## R&TD objectives

- (1) To design conceptually the MASS instrument on Small Satellite
- (2) To study the science capabilities/error budget for the MASS and perform simulation of its expected science impact
- (3) To demonstrate the key laser pixel metrology calibration technique using a large state-of-the-art 150Mpix sCMOS detector



# Laser interferometer metrology for calibration

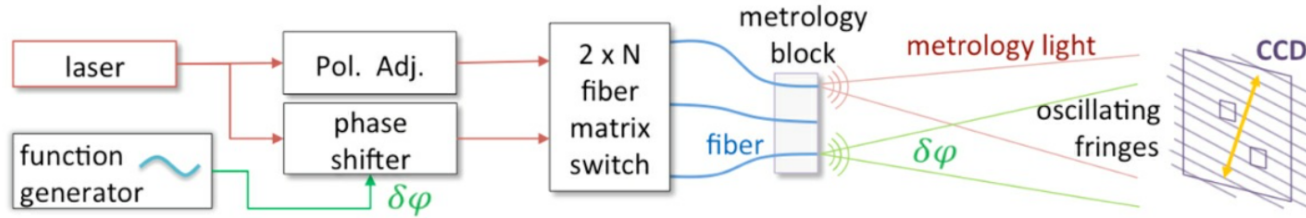
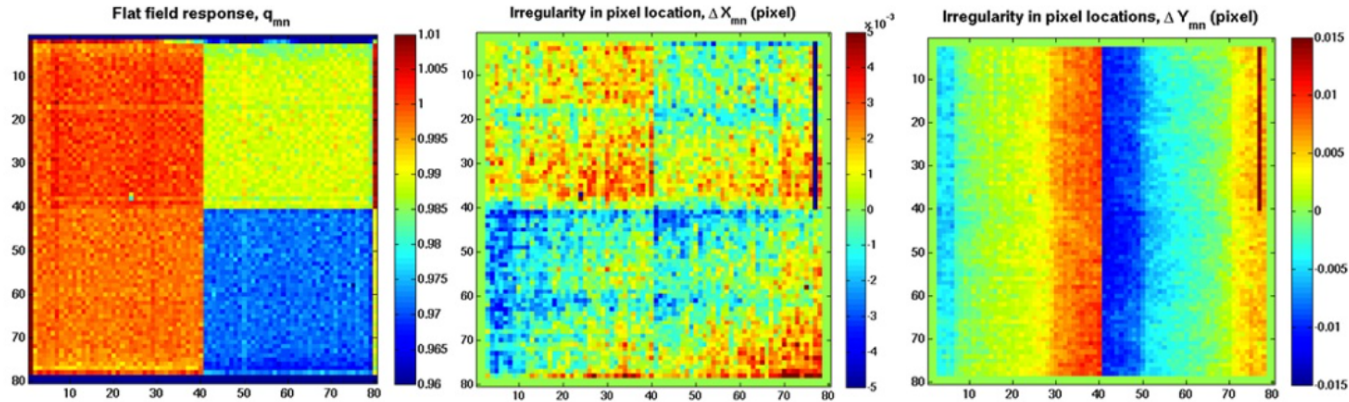
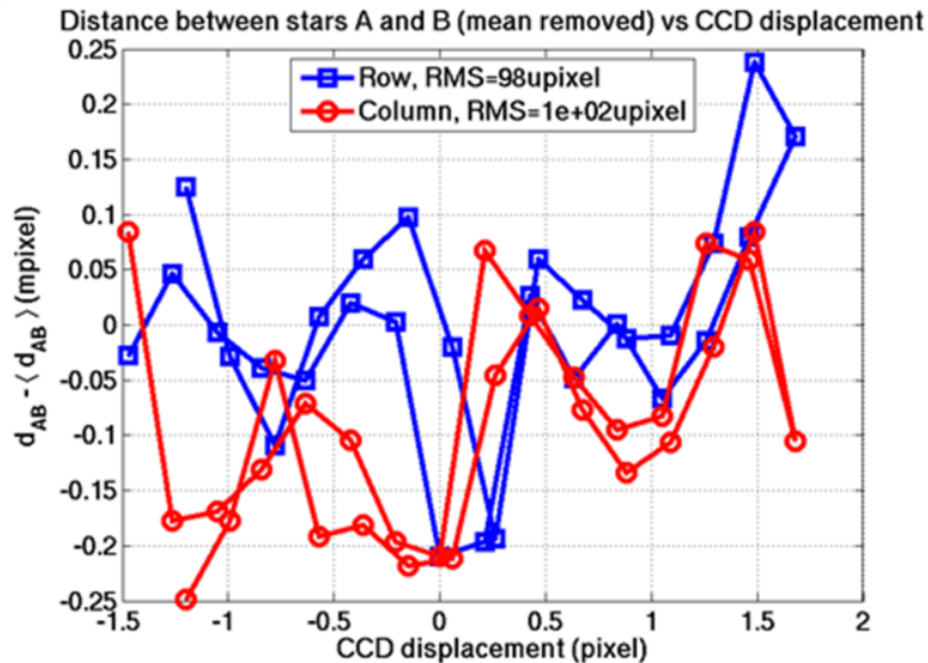
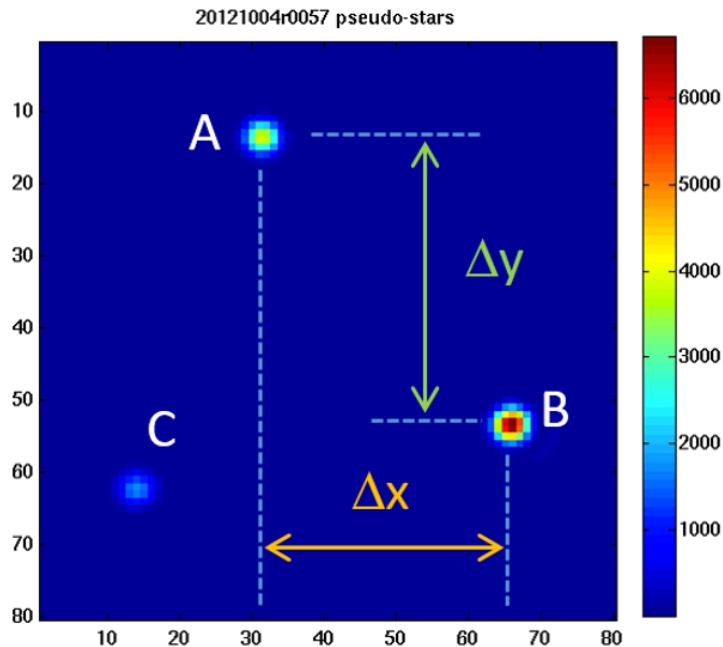


Figure 3. Laser metrology for calibrating irregularities in pixel locations.



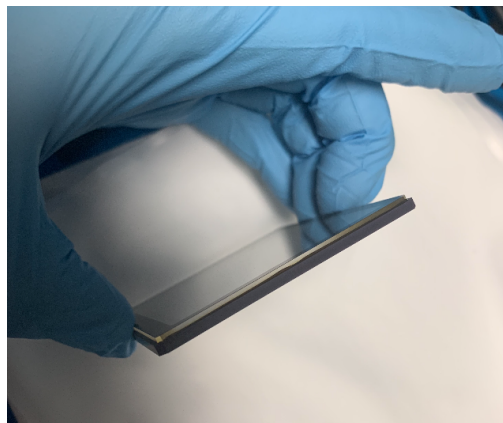




(Left) An astrometric test measures the consistency of inter-star distances on the focal plane as the line of sight is changed. (Right) Results of an astrometric test: centroid distance between pseudo-stars A and B in row and column directions, with mean removed, versus the displacement of the CCD.



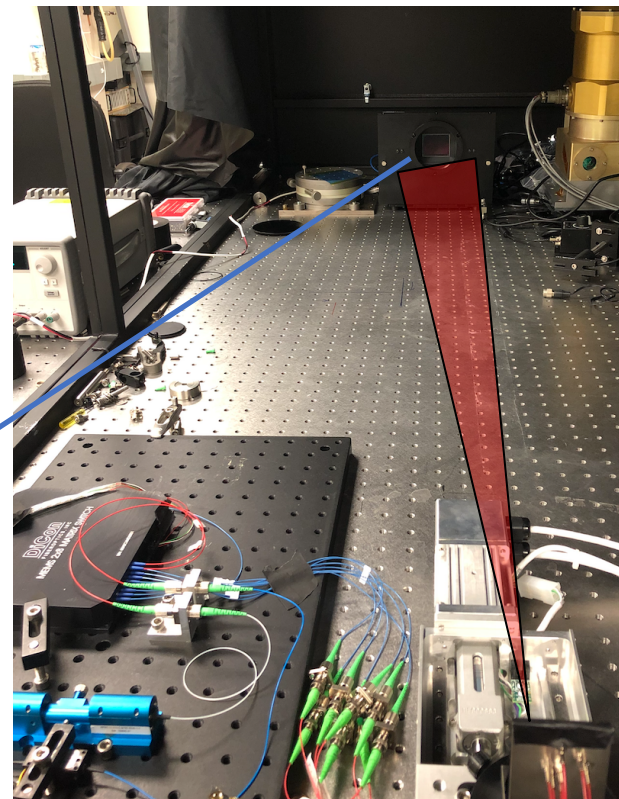
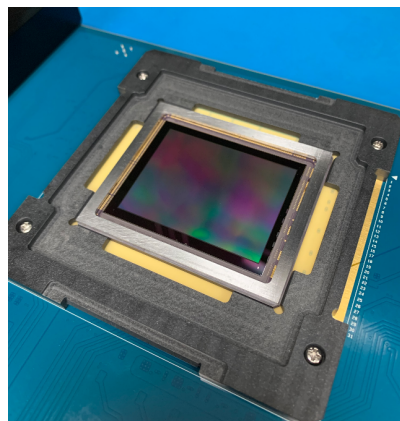
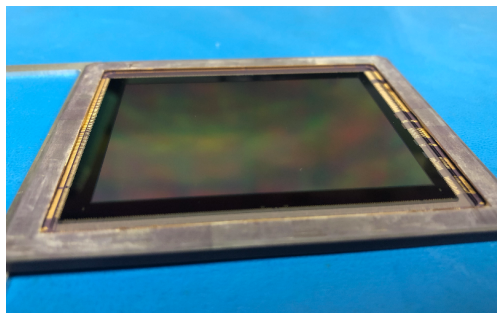
## Large detector calibration



Sony IMX411

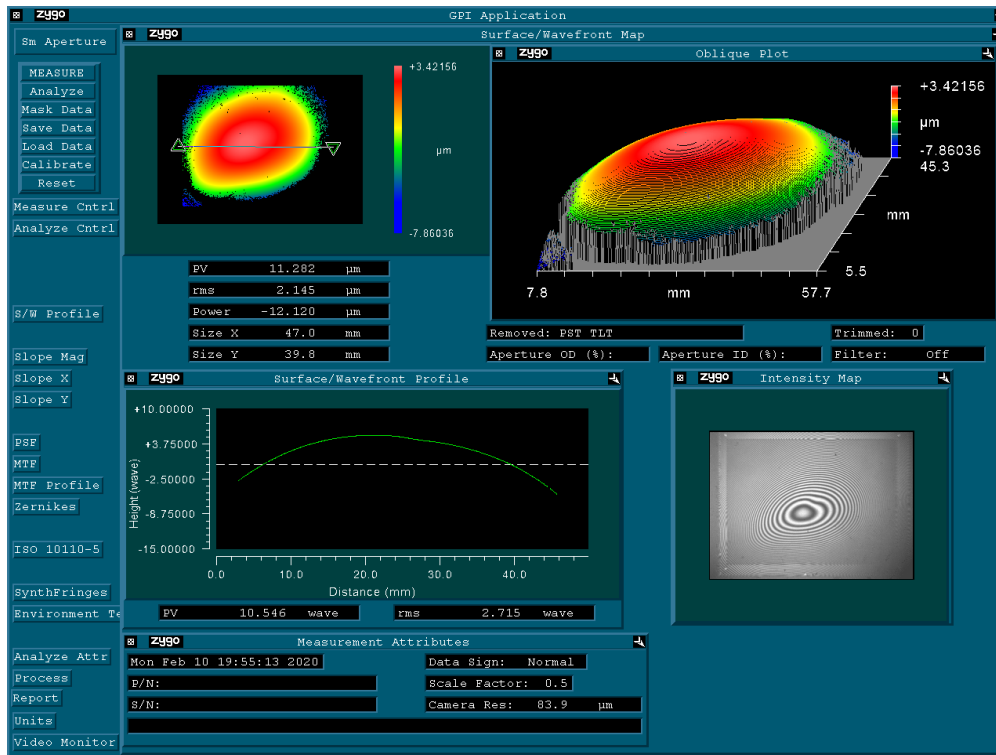
~150 Mpix,  $3.76 \mu\text{m}$ ,  
backside illuminated,  
<  $2e^-$  read noise

Glass window is removed





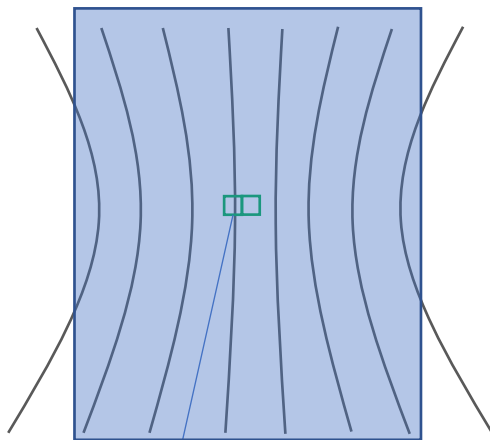
# Chip mount distortion measurement



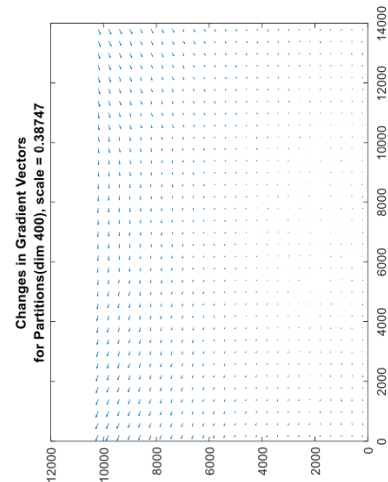
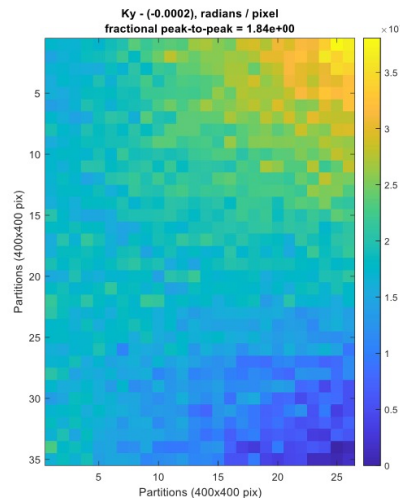
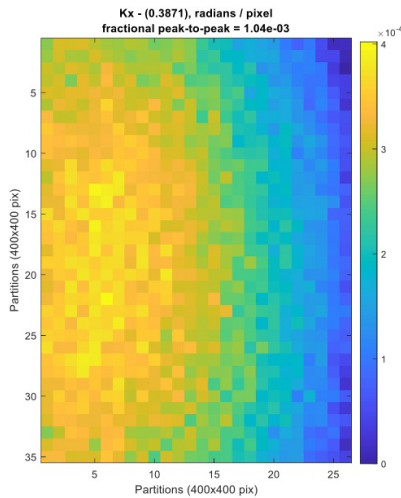
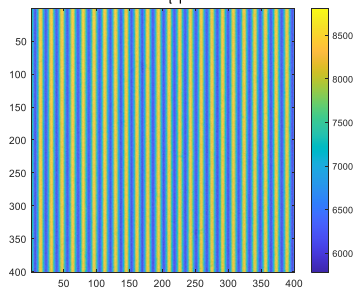
~11 micron distortion (3 pixel over 12000 pixels ~  $1.6 \times 10^{-4}$ ) in z-direction has to be corrected for the final metrology analysis



## Non-linear fringe effects



frame\_0.it



Full frame data were taken  $\sim(10,000 \times 14,400)$

Partitioned into  $400 \times 400$  regions ( $26 \times 35$  measurements of  $K$ )

The analysis of the local  $K$  vector revealed the non-linear effects.

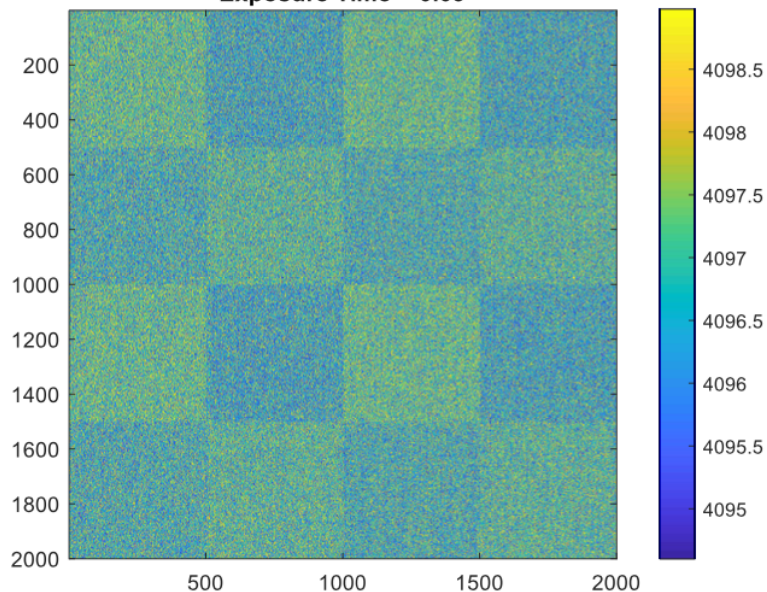
$$\vec{K} = \frac{2\pi}{\lambda} \vec{v}OPD$$

The effect is at  $10^{-4}$  level across the chip, which has to be taken into account for the pixel irregularity determination.

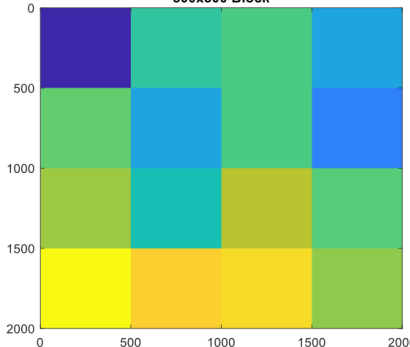


# Birthmark Features

Raw Mean Dark Frame  
Exposure Time = 0.09

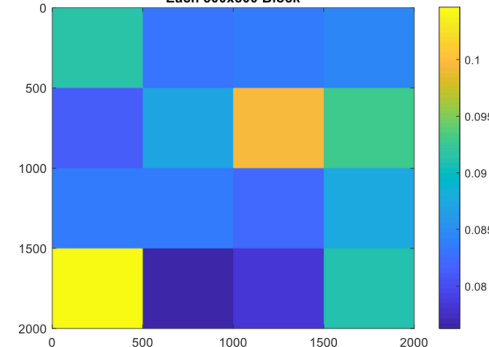


Bias for Each  
500x500 Block

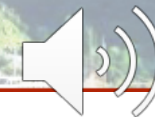


K-gain = 0.678

Dark Current for Each  
500x500 Block



Dark frame image showed a 500x500 pixel block pattern



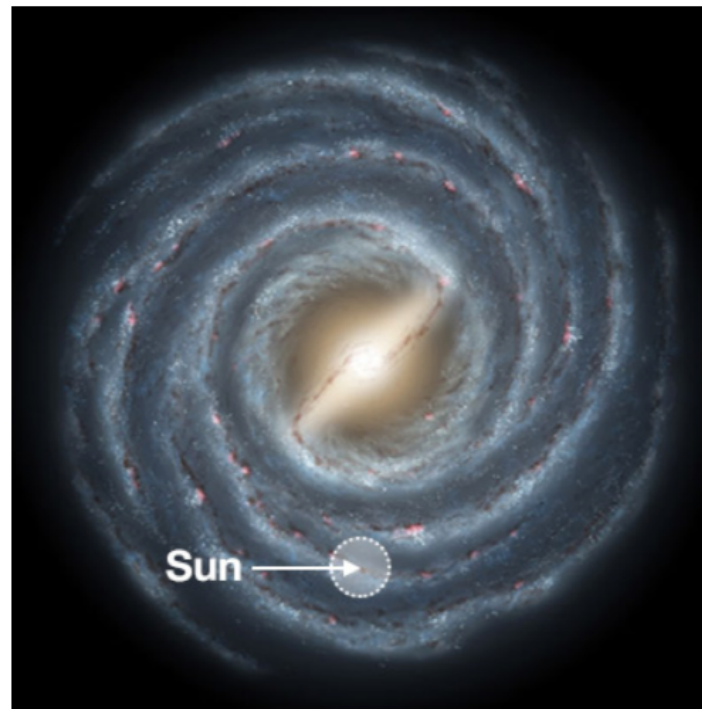
## Summary

### Significance

- The concept of the MASS mission was submitted to 2019 Astrophysics Explorers Mission of Opportunity.
- “MASS will be a game-changer for NASA’s goal of exploring Earth analogs by pioneering space base astrometric detection and characterization of exoplanet, and the mass measurement of the identified planets is essential for interpretation of spectra for future direct-imaging missions.” -reviewer
- “pixel metrology demonstration with a large size detector is required” -reviewer

### Next Steps

- The task was halted during the mandatory lab closure due to COVID-19
- The unfinished activities are (1) taking the metrology data in various experimental set up (2) data analysis, (3) publication of the results



The sun and our solar system in relation to the Milky Way galaxy. The white circle indicates the area where the majority of exoplanets have been found with current telescopes. Credit: NASA/JPL-Caltech/T. Pyle